

$$km(x, \beta) := \beta_1 + \frac{\beta_8}{x + \beta_2 - \frac{\beta_9}{x + \beta_3 - \frac{\beta_{10}}{x + \beta_4 - \frac{\beta_{11}}{x + \beta_5 - \frac{\beta_{12}}{x + \beta_6 - \frac{\beta_{13}}{x + \beta_7}}}}}$$

$$\beta := \begin{bmatrix} -16702.7193446 \\ 124.751615769 \\ -208.000512345 \\ -16.7911347830 \\ -4.9542542218810^{-2} \\ -8.76777990888 \\ -1.33046986462 \\ 4191546.15004 \\ 4221.26725909 \\ 5327.77204957 \\ 1.34644054551 \\ 4.17574113874 \\ 0.274345957759 \end{bmatrix}$$

```
plot(data, char, size, clr) := for k ∈ 1..rows(data)
    [ r3_k := char r4_k := size r5_k := clr ]
    augment(data, r3, r4, r5)
```

```
data :=
    0      60000
    0.5    35462
    1      30747
    2.1    26217
    3.6    22781
    5.6    19921
    9.7    16453
    14.7   13766
    22.3   11115
    32.4   8654
    50.2   5563
    75.5   2423
    101.325 0

data := plot(data, "o", 5, "black")

{ XY := data 1..13 1..2
  [ X := col(XY, 1) Y := col(XY, 2) ]

Absolute residuals

Δ := Δ := 0
for j ∈ 1..rows(XY)
    Δ_j := eval(Y_j - km(X_j, β))
res := augment(X, Y, round(Δ, 12))
stack(["kPa" "alt[m]" "Δm"], res)
```

$$k(x) := \begin{cases} 1 & \text{if } (0 \leq x) \wedge (x \leq 101.325) \\ "" & \text{otherwise} \end{cases}$$

Δm := col(Δ, 3)

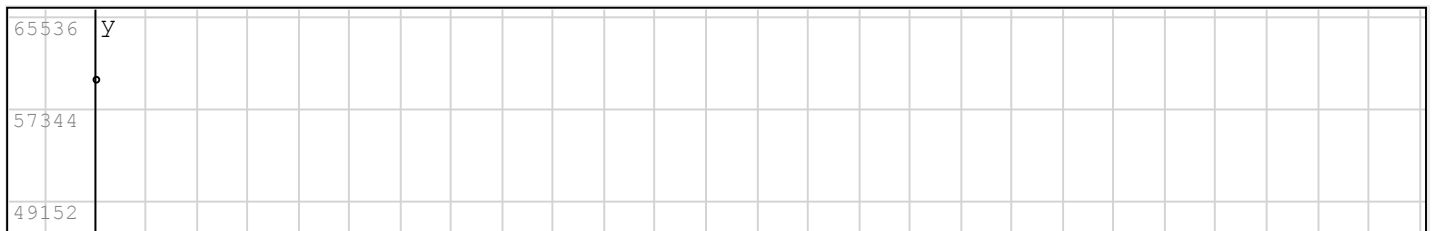
Δ =

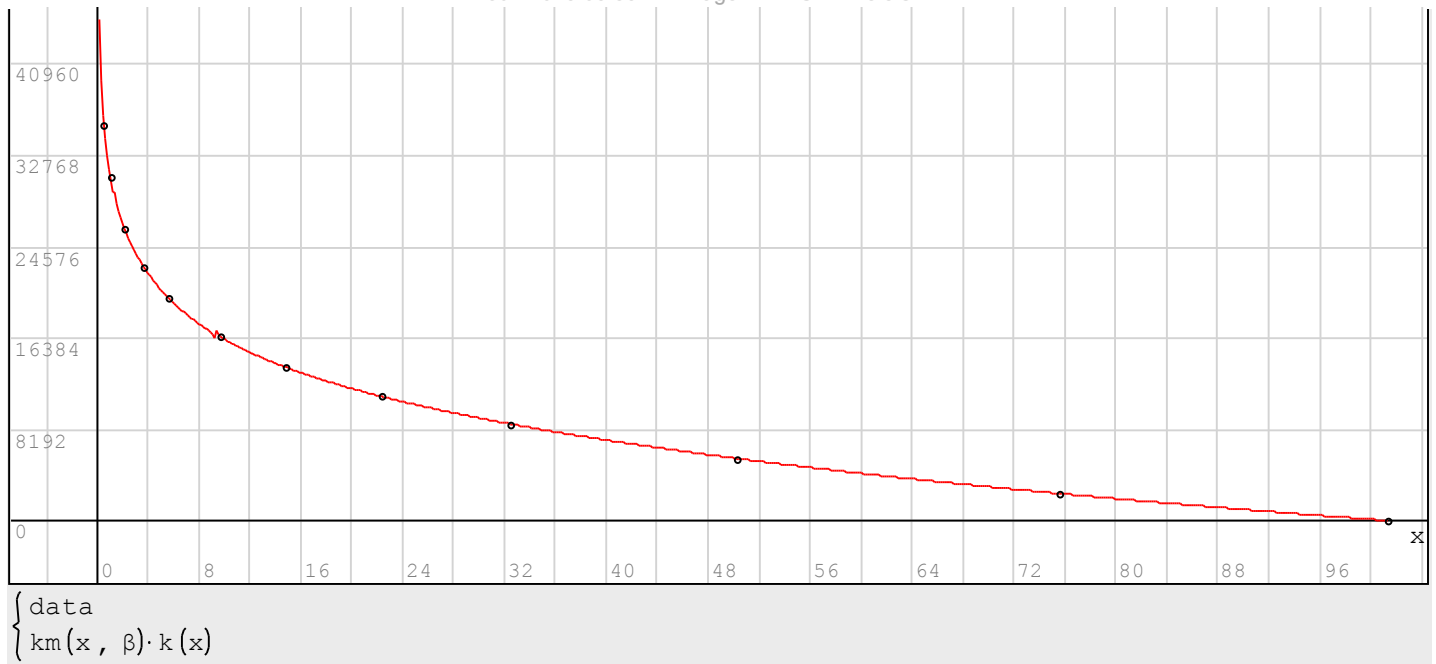
"kPa"	"alt[m]"	"Δm"
0	60000	0
0.5	35462	0
1	30747	0
2.1	26217	0
3.6	22781	0
5.6	19921	0
9.7	16453	0
14.7	13766	0
22.3	11115	0
32.4	8654	0
50.2	5563	0
75.5	2423	0
101.325	0	0

Δm =

"Δm"
0.000000
0.000000
0.000000
0.000000
0.000000
0.000000
0.000000
0.000000
0.000000
0.000000
0.000000
0.000000
0.000000
0.000000
0.000000

altitude [m] vs measured pressure [kPa] ... i.e: InverseNASA(h)





<https://www.grc.nasa.gov/WWW/K-12/airplane/atmosmet.html>



"<https://www.grc.nasa.gov/WWW/K-12/airplane/atmosmet.html>"

NASA atmos-airplane <https://www.grc.nasa.gov/WWW/K-12/airplane/atmosmet.html>